

TOLIMAN JNEXT



Jet Propulsion Laboratory
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Searching for earth analogs around the nearest star system

TOLIMAN JNEXT

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2. NASA Ames Research Center
3. IPAC Caltech
4. University of Sydney
5. University of Michigan
6. University of New Mexico
7. Paris Observatory

Exoplanet Exploration Strategy

MOTIVATED BY NASA ASTROPHYSICS ROADMAPS

Priority Area: Pathways to Habitable Worlds

We are on a path to exploring worlds resembling Earth and answering the question: “Are we alone?” The task for the next decades will be finding the easiest of such planets to characterize, and then studying them in detail, searching for signatures of life.



Astrophysics Decadal Survey (2020)

Specific tasks:

1) Finding the easiest targets to characterize and study Earth analogs in detail.

How?

List of best stars to study Earth analogs

- Distance
- Stellar type

Then, search for planets around those stars

Specific tasks:

2) Then study them in detail searching for signatures of life.

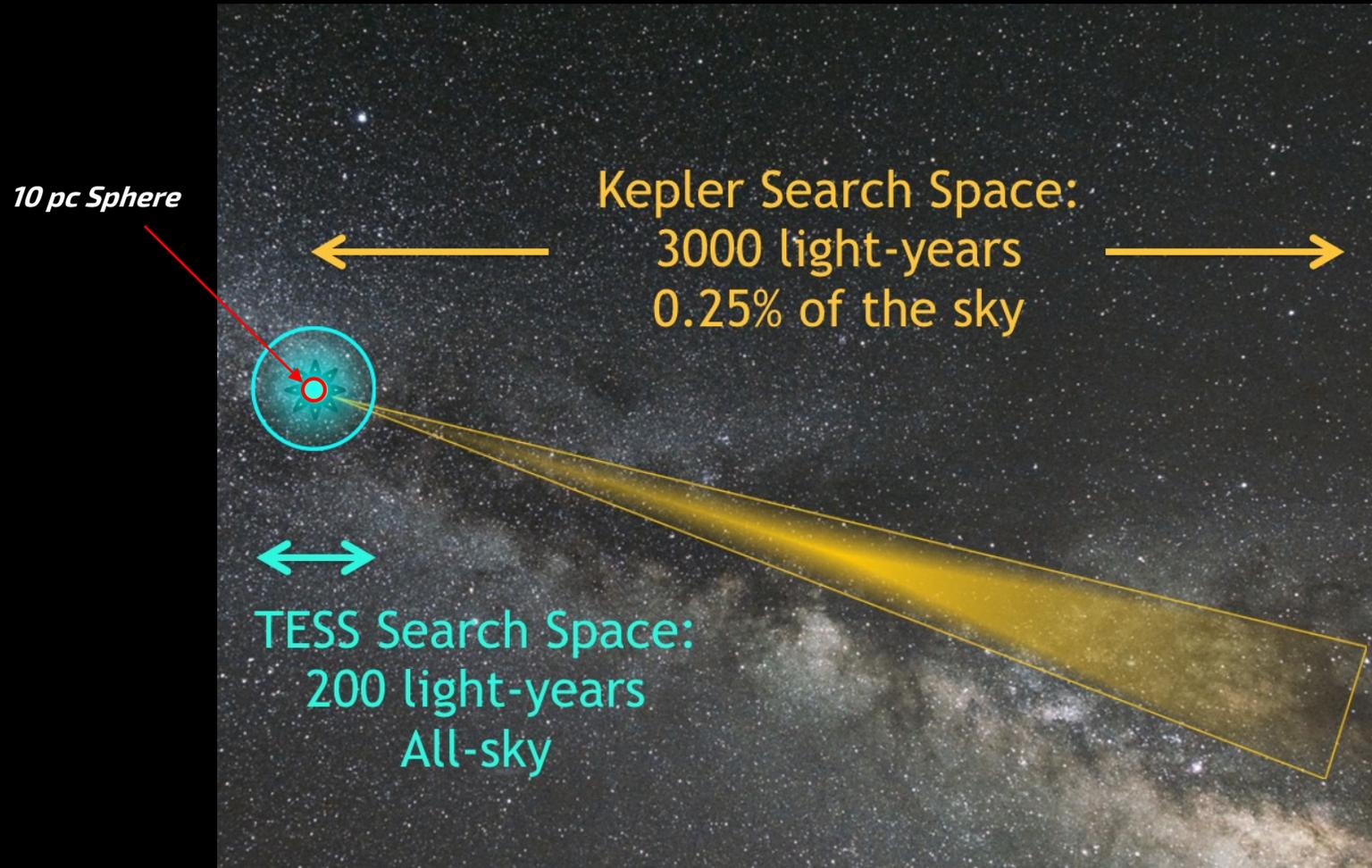
Key parameter harder to measure by UV/O/IR mission:

- Mass determination

Why measure masses?

- Distinguish terrestrial rocky planets from giant planets
- Determine system inclination, confirm RV.
- Facilitate retrieval of atmospheres’ chemical species
- Need to complete the “extrapolation region”
- HabEx, LUVOIR, and Starshade have limited ability to infer masses

From demographics to habitability studies



The TESS search space compared to that of the Kepler Mission. Image Credit: Zach Berta-Thompson

SCIENCE

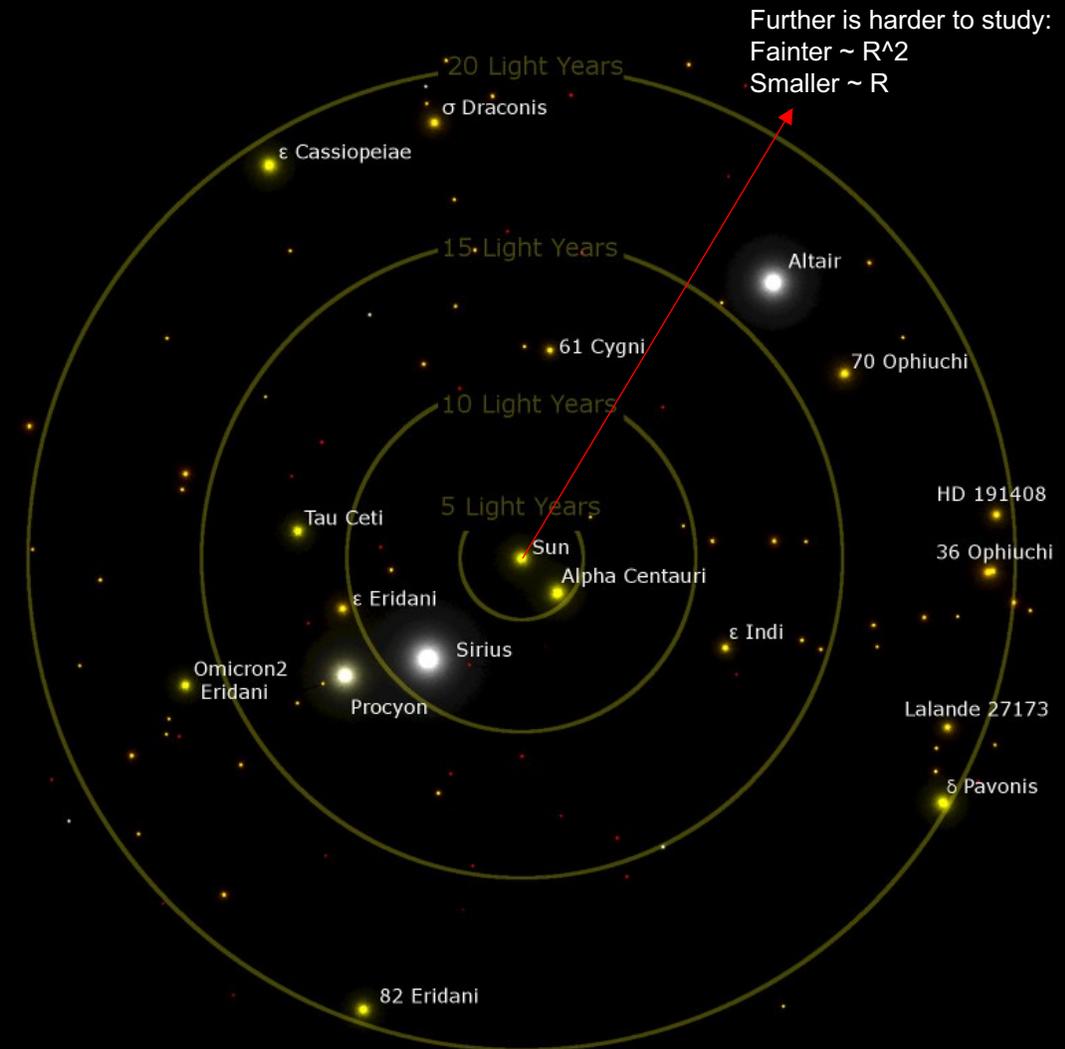
1) Finding the easiest targets

Our stellar neighborhood, few targets, each one is valuable

- Only 67 sun-like stars are within 10 pc radius, 41 of those are binary systems => **We cannot ignore binaries**
- *Alpha Centauri A & B* is three times closer than the next sun analog *Tau Ceti*. Planets around α Cen A&B are:
 - ~9x brighter
 - ~3x greater resolutionthan around next Sun analog, Tau Ceti

Observing an earth analog at 10 pc with a 6 m UV/O/IR is equivalent to observe aCen A&B with a 0.6 m aperture small sat

- Time critical because informs the design of an exoplanet flagship (Kepler, GAIA, TESS, **cannot** observe aCen A&B)



“Alpha Centauri system, if not for the fact that it is a binary, would easily be the best target for direct imaging searches for planets”

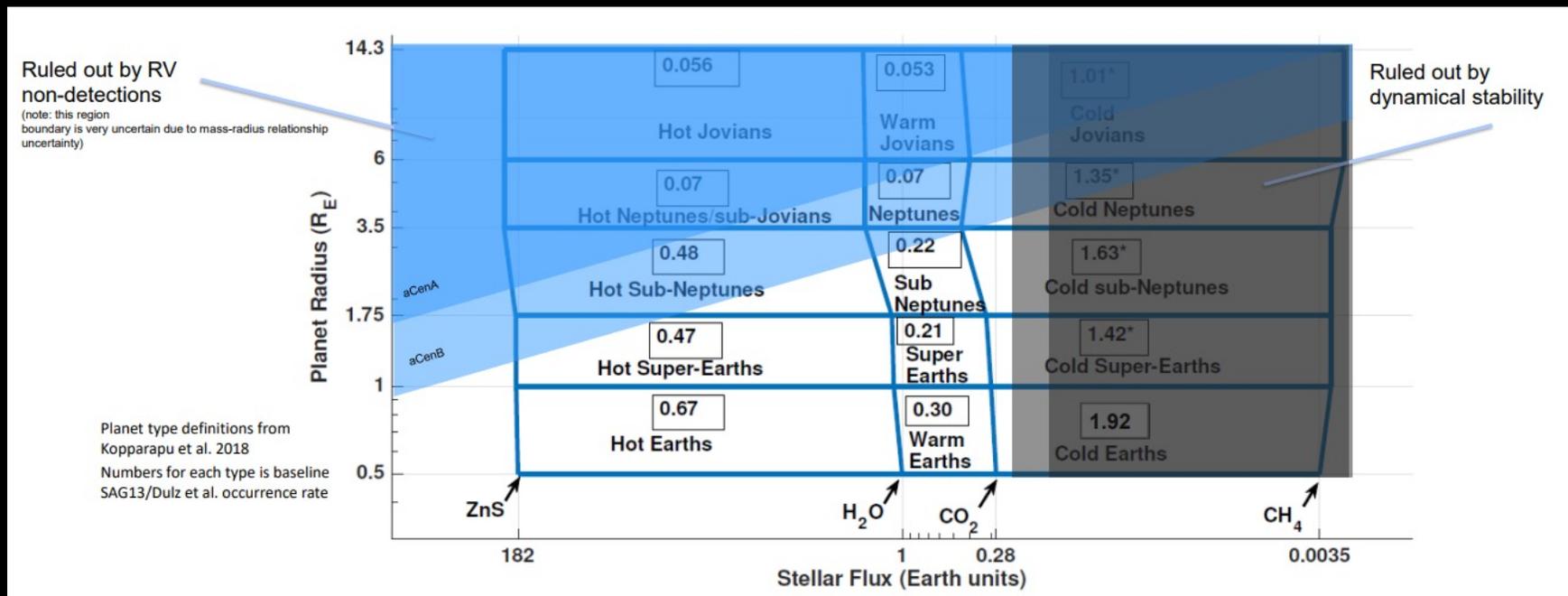
--HabEx Final report

aCen A&B state-of-the-art

Binarity and stellar noise limits instrumentation sensitivity to detect earth-analogs around the best target!

- Stability limits allows planets beyond the H.Z.
- *Earth analogs remain possible (and likely) around both stars*
- *R.V. limits have ruled out Neptunes or larger planets*

There is a knowledge gap of the entire H.Z. of aCen A&B because instrumental limitations



Belikov et. al, 2022 SAG13 update at (Exoplanets IV)

APPROACH AND TECHNOLOGY

Using the binarity as a solution

This concept uses relative astrometry between binaries to detect planets and measure their masses

Challenge: The signal is $\sim 1 \mu\text{as}$, optical distortion term dominates. State-of-the-art is about 25 times less accurate (Hubble, GAIA, Gravity $\sim 25 \mu\text{as}$)

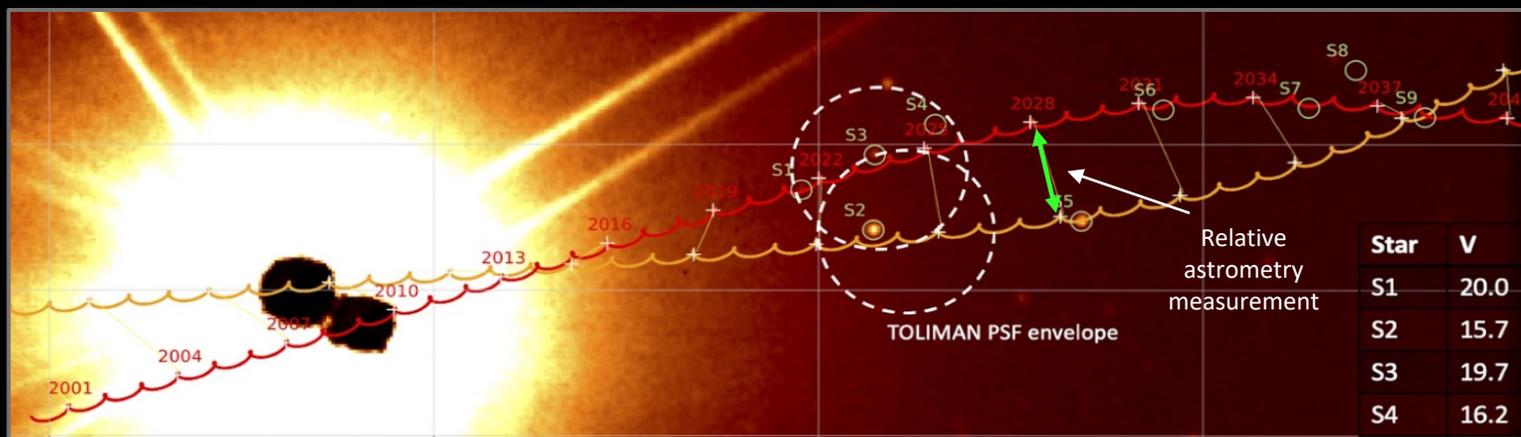


Image of the $\alpha\text{Cen A\&B}$ field taken from the ESO's NTT 4.0m

Courtesy: P. Kervella.

Opportunity that $\alpha\text{Cen A\&B}$ offers:

- **Proximity:** An Earth analog around $\alpha\text{Cen A\&B}$ will exhibit the **highest astrometric signal possible** of $\sim 3 \mu\text{as}$
- **Brightness:** A 9cm aperture collects enough photons from $\alpha\text{Cen A\&B}$ to reach $1 \mu\text{as}$ photon noise in 6hrs!

APPROACH AND TECHNOLOGY

What limits the astrometric detection of earth analogs?

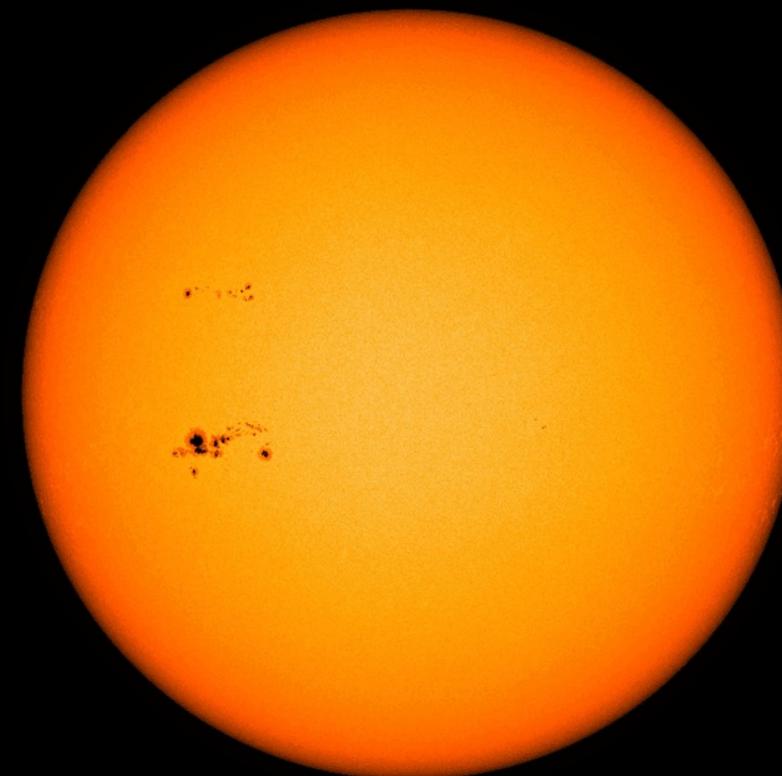
Photon noise

A 9cm aperture collects enough photons from α Cen A&B to reach $1\mu\text{as}$ photon noise in 6hrs!

Astrophysical noise is benign (in contrast to EPRV)

- Sun-like stars at 10pc viewed from equator = $0.087\mu\text{as}$ jitter Marakov et al 2009 (ApJ 707, L73)
- Similar study in 2011 is consistent = $0.07\mu\text{as}$ RMS, $0.2\mu\text{as}$ PV Lagrange et al 2011 (A&A 528, L9)
- Absolute astrometric Jitter from solar data = $0.52\mu\text{AU}$ (0.11mR) jitter Marakov et al 2010 (ApJ 717, 1202)
- Signal earth-like planet around the sun: $0.46\text{mR} = 2.17\mu\text{AU} = 0.3\mu\text{as}$ @ 10pc
- Peer reviewed literature agrees on a stellar astrometry jitter $\sim <0.1\mu\text{as}$

Several studies shows that stellar jitter is NOT a showstopper for astrometric detection of earth-analogs.



SDO/HMI Quiescent Continuum: 20140201_000000

Sun spot AR1967. Animation: SDO/NASA.

APPROACH AND TECHNOLOGY

What limits the astrometric detection of earth analogs?

Instrument

Optical distortion

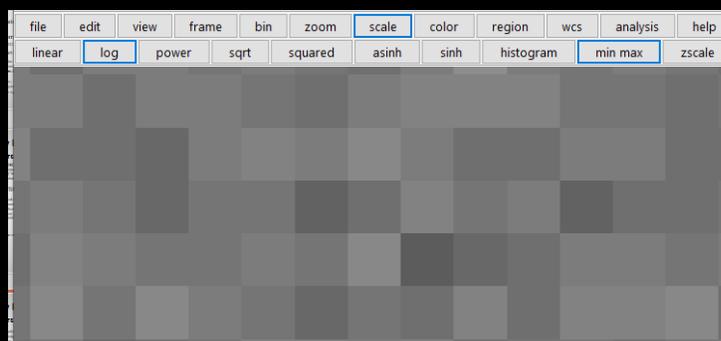
- The image or magnification changes in the entire or part of the field of view.
- Static distortion is OK
- Dynamic distortion changes the scale of every epoch

Detector errors

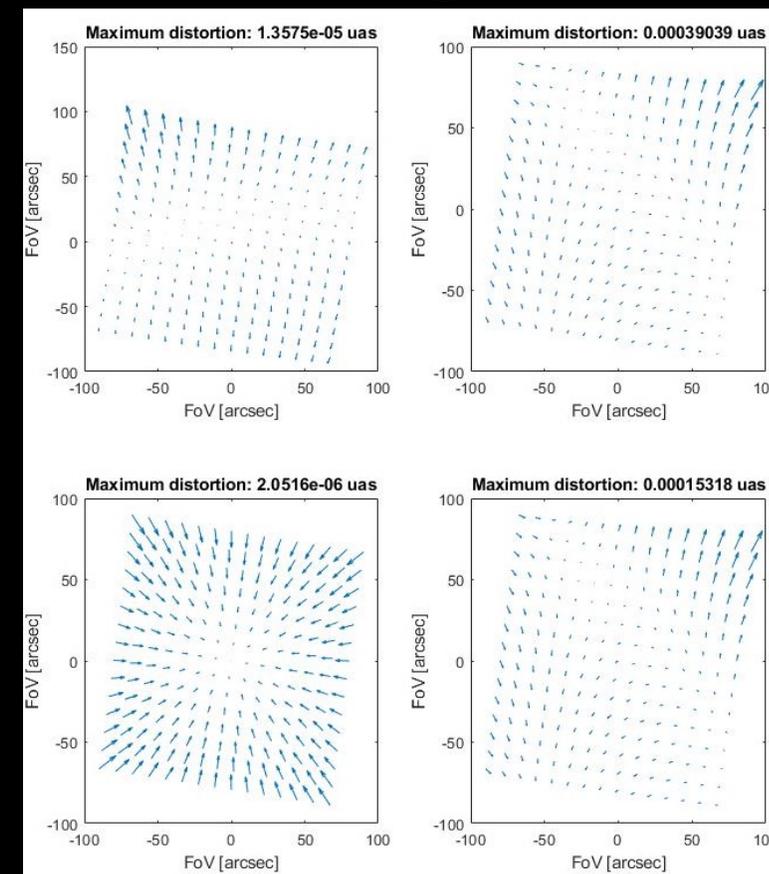
=>If the PSF moves to a differ pixel the response is different.

- Pseudo static: Inter and intra pixel QE changes, pixels position, pixel gaps, circuitry for front illuminated CMOS. => “Flat field”
- Dynamic: Thermal expansion, brighter-fatter effect, detector bending, etc

This also applies to photometric observations, Kepler, Tess, Plato, etc



Typical flat field error under uniform illumination. This is 3.76
um pixel BSI CMOS



Sample of modal of distortion of the Habex Work
Horse camera (4 modes of 121 shown)
Astrometry exoplanet detection using the Habex Work
Horse Camera Bendek et al. Proc SPIE 11823

The diffractive pupil approach

Single and binary stars optimizations: Solve distortion and camera pixel errors

Absolute stellar astrometry

Astrometry reference are the background stars with enough SNR resulting in:

- Larger telescope
- Wide FoV
- Solve target v/s bkg dynamic range problem
- Guyon, O., and Coauthors, 2012: High-precision Astrometry with a Diffractive Pupil Telescope. *The Astrophysical Journal Supplement Series*, **200**, 11.]
- Bendek, E., et al., 2013: Laboratory demonstration of astrometric compensation using a diffractive pupil," *PASP*, Vol. 125, No. 932, pp. 1212-1225.
- Bendek, E., et al, 2017: Results of the astrometry and direct imaging testbed for exoplanet detection," *Proc. SPIE*, 10400.

Narrow field relative astrometry

Astrometry reference is binary companion:

-> Smaller telescope

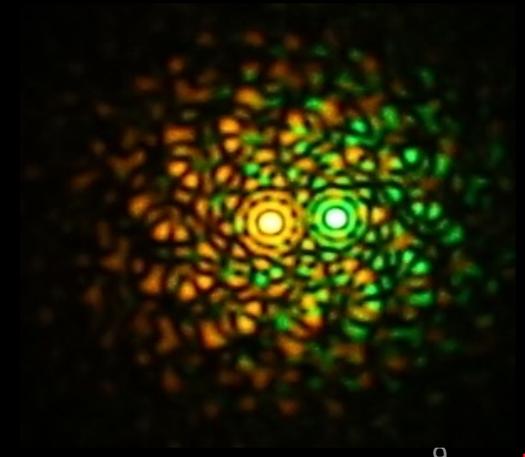
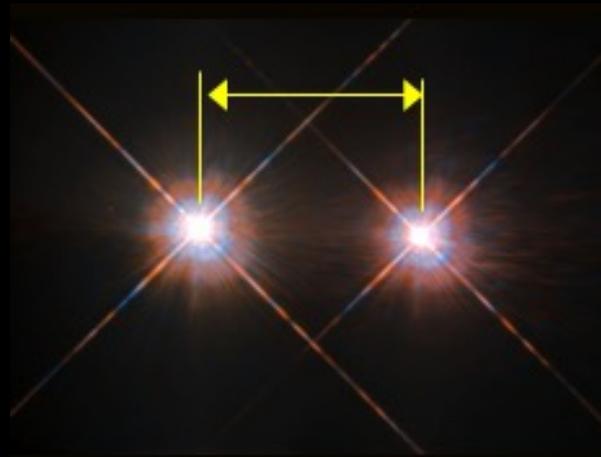
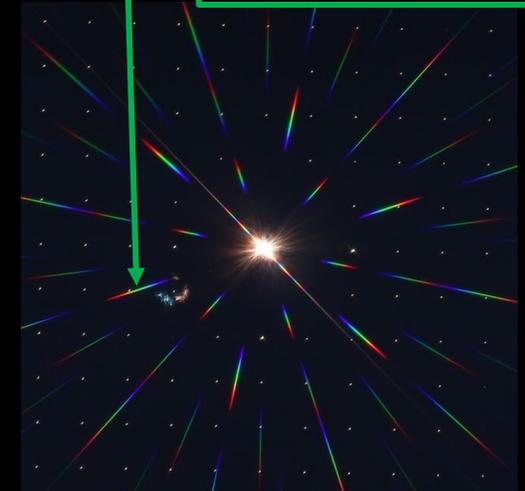
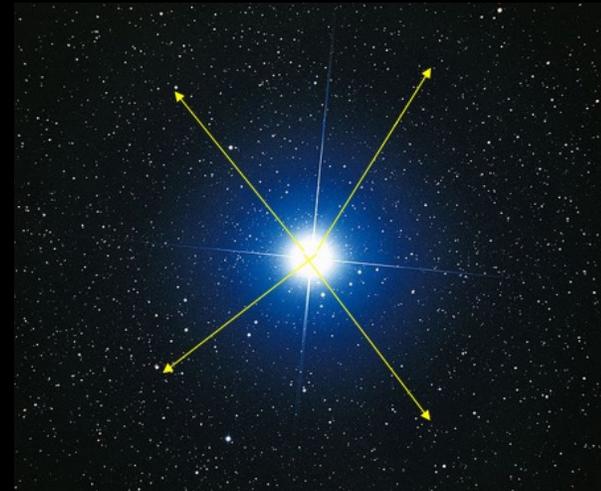
Narrow field of view to capture the binary

-> Simple telescope

- Bendek, E., and Coauthors, 2018: Precision astrometry mission for exoplanet detection around binary stars, *Proc. SPIE*, 10698.
- Tuthill, P., and Coauthors, 2018: The TOLIMAN space telescope. 107011J.

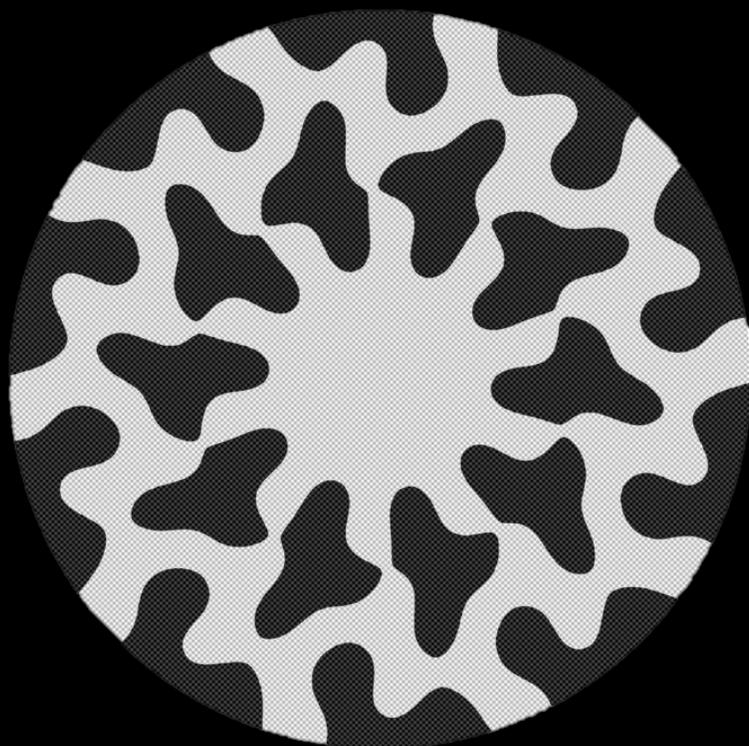
See **Catalina Flores Quintana poster** to learn how to reduce the diffraction spikes data

See **Matt Noyes poster** about our picometer controller active light source



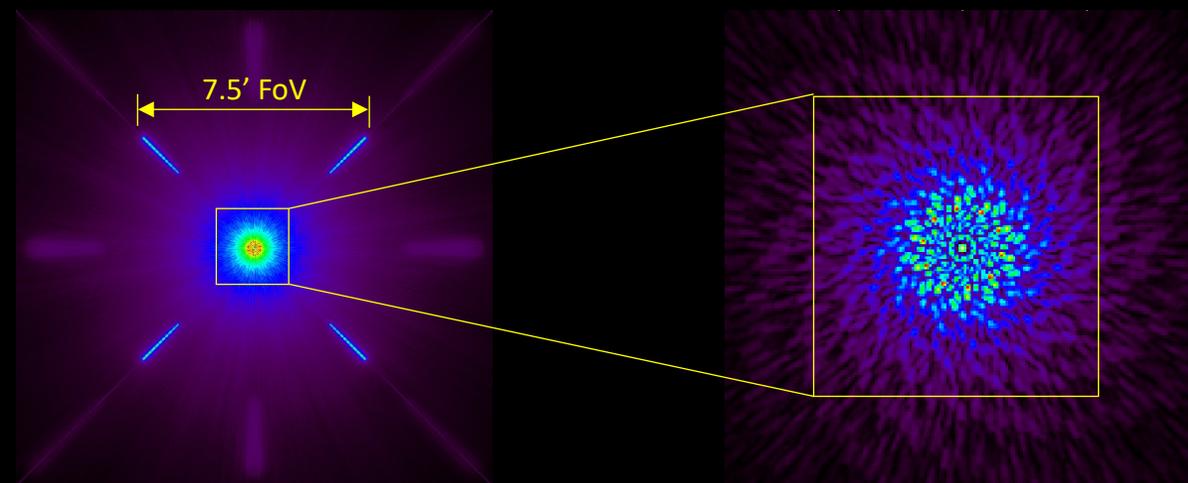
Diffraction Pupil for binaries

Pupil plane (Primary mirror)



Diffraction pupil topography
 $\lambda/4$ height difference between black and white

Focal plane Point Spread Function (PSF)



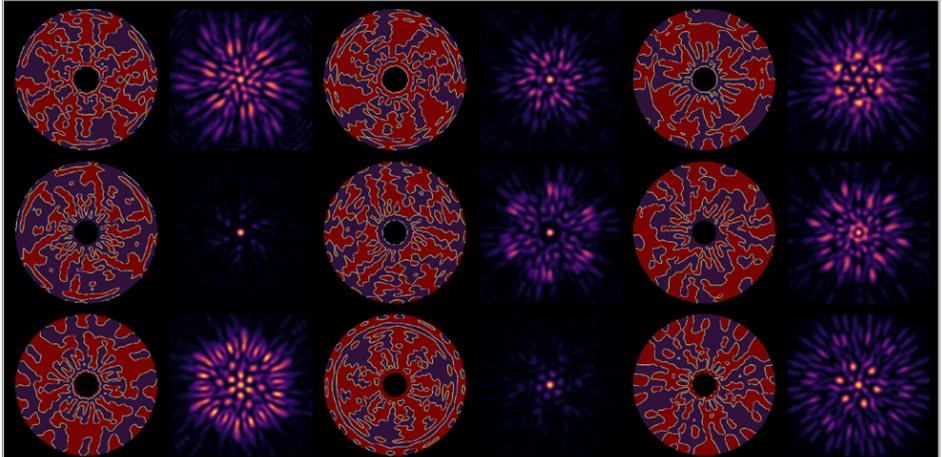
Diffraction Pupil for binaries optimization

- Univ. of Sydney has developed advance code to explore the DP shape optimization
- See Louis Desdoigts Poster #45: Optical Design, Analysis & Calibration using δ Lux
- TOLIMAN development at **Univ. of Sydney** led by **Prof. Peter Tuthill** funded by breakthrough initiatives.

 THE UNIVERSITY OF SYDNEY

Design Methods: Gradient Energy Optimisation

- CLIMB: Map latent continuous basis to binary mask in a continuous manner – Latent basis symmetries mapped onto binary mask
- Loss function: Tweaked gradient energy metric to maximize PSF gradients distributed within 10 lam/D
- Optical Model: Current proposed optical configuration, full 100nm chromatic bandpass



Louis Desdoigts Poster #45: Optical Design, Analysis & Calibration using δ Lux

JNEXT TOLIMAN Project goals:

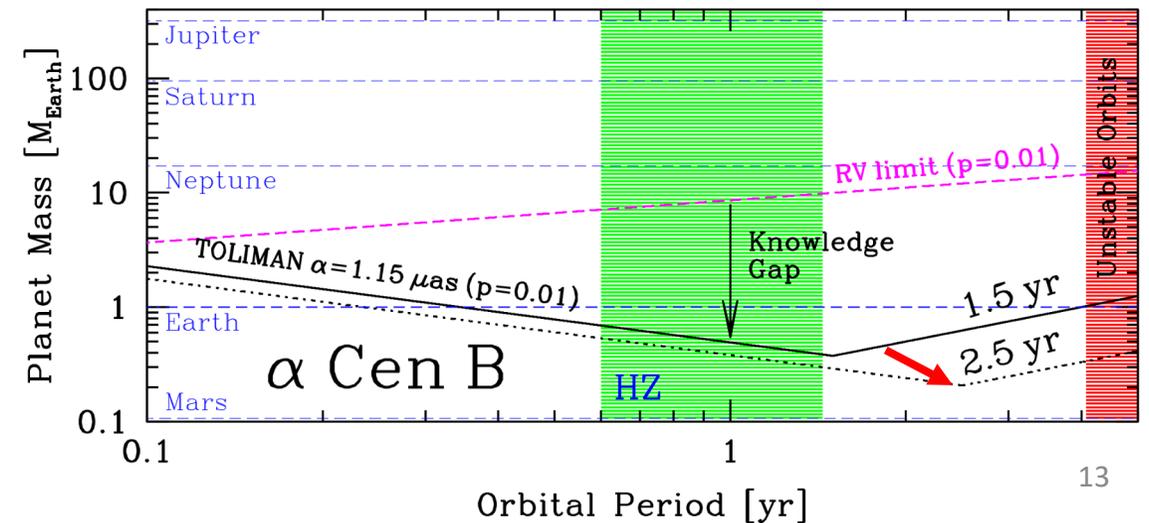
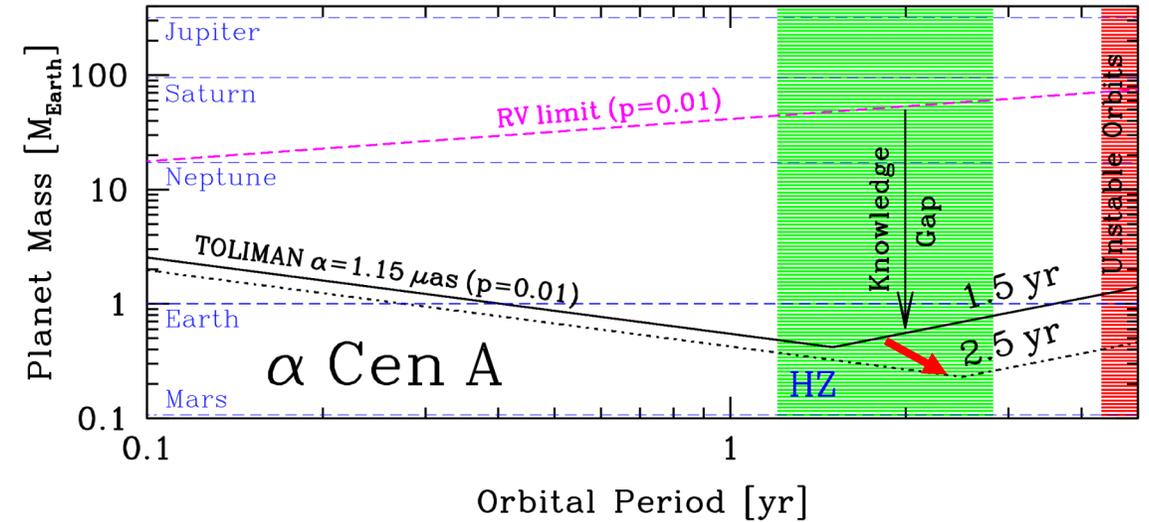
Technology: Demonstrate the diffractive pupil technology in preparation for larger NASA missions that can survey the nearest binaries.

Science: Survey the entire habitable zones of Alpha Centauri A and B with enough sensitivity to detect and measure masses or earth analogs

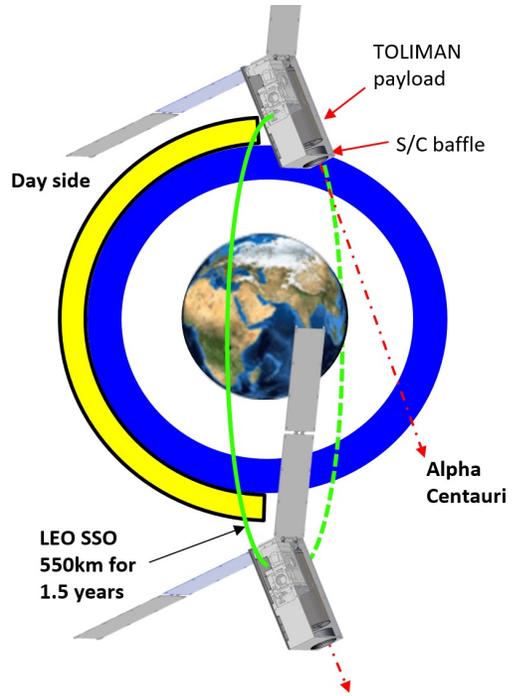
Discovering an earth analog around Alpha Centauri, our closest star system, would be a major scientific breakthrough and will herald the search for life and biosignatures.

System Trades: Mission Duration

- **Mission duration** drives sensitivity, not number of targets
- Minimum duration: 1.5 years, ideal 2.5 years
- **Options:** gradual increase in mission duration. S/C is designed for 5 yrs in LEO SSO.
- **Trade:** Higher cost, more planets, longer period, and enables detection of Mars analogs



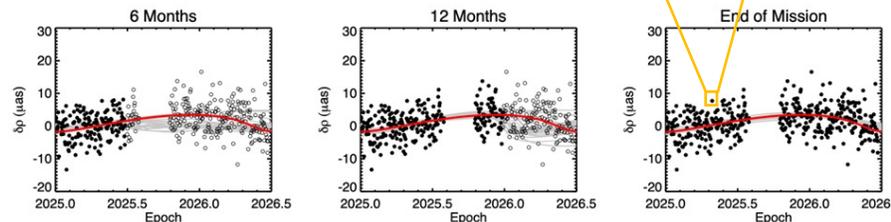
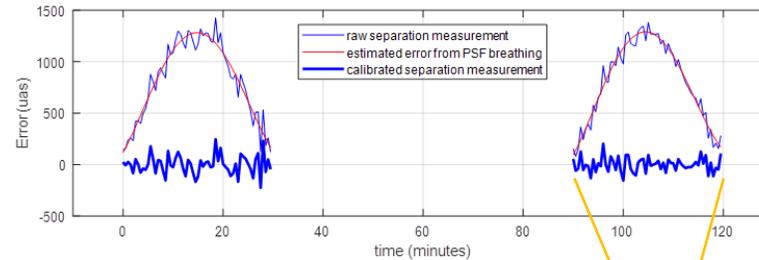
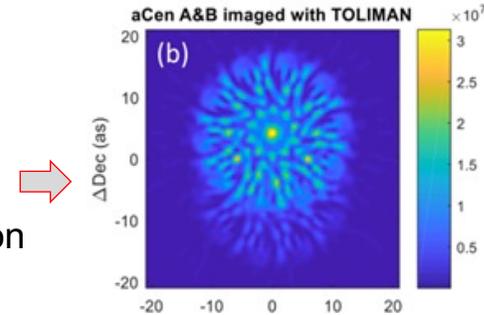
The TOLIMAN mission in a nutshell



- TOLIMAN uses a BCT 6U CubeSat.
- The instrument is a 9cm aperture Cassegrain telescope equipped with the DP
- The Mission duration is 1.5 years

TOLIMAN Data:

- 1 observation/orbit
- 9,000 frames, 0.2 s each
- PSF contains plate scale calibration



Orbital fit provides planets, masses, orbit

L0
Raw data
(onboard)

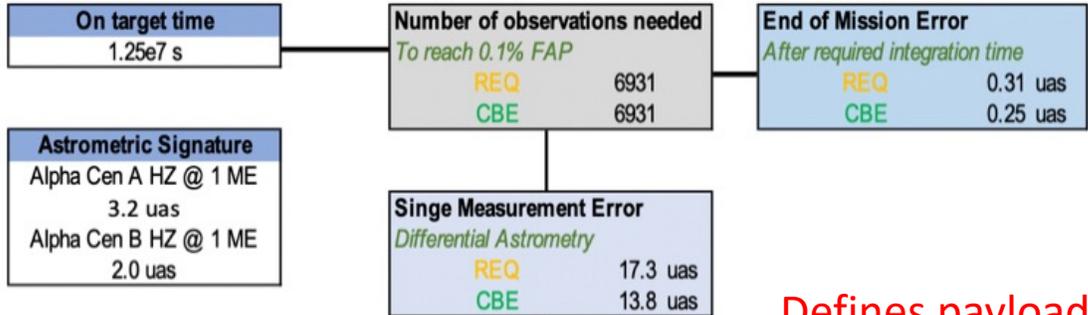
L1
Calibrated
astrometry
vectors
(Ground)

L2
Astrometry
observation
(Ground)

L3
Planet and
orbital data
(Ground)

Mission Error Budget

Target	A Cen AB	
V Mag A B	-0.01	1.33
Telescope Diam.	9 cm	
Efficiency	0.46	
Wavelength	550 nm	
Bandwidth	110 nm	
PSF Sampling	0.33 lam/D	
Integ. time / obs.	30 min	



Defines payload requirements

S/C Pointing <i>PSF smearing from LOS change</i>	REQ 5.2 uas CBE 3.9 uas
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Photon & Detector Noise <i>Differential Astrometry</i>	REQ 10.0 uas CBE 9.0 uas
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Alpha Cen A <i>Photon Noise, Read Noise and Dark Current, Flat Field Knowledge, Gamma factor 2 (see text)</i>	REQ 5.2 uas CBE 4.9 uas
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Alpha Cen B <i>Photon Noise, Read Noise and Dark Current, Flat Field Knowledge, Gamma factor 2 (see text)</i>	REQ 8.5 uas CBE 7.6 uas
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Focal Plane Error <i>Pixel location knowledge</i>	REQ 4.6 uas CBE 3.2 uas
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Pixel calibration & centroiding <i>Ground calibration of sensor</i>	REQ 3.5 uas CBE 3.0 uas
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Thermal changes <i>Since calibration</i>	REQ 3.0 uas CBE 1.2 uas
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Field Distortion Error <i>Telescope and OTA</i>	REQ 8.0 uas CBE 6.2 uas
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Diffractive Pupil Calibration <i>After on-sky calibration</i>	REQ 4.5 uas CBE 4.0 uas
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Platescale Calibration <i>After on-sky plate scale calibration</i>	REQ 5.2 uas CBE 3.8 uas
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Diffractive Pupil Changes <i>Therm-Mech changes in diff. pupil</i>	REQ 4.0 uas CBE 2.8 uas
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Astrophysical Error <i>Target and background stars</i>	REQ 9.5 uas CBE 6.8 uas
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Astrometric jitter AB <i>AB jitter on relevant timescales</i>	REQ 1.0 uas CBE 0.8 uas
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Background stars <i>After calibration of known objects</i>	REQ 5.0 uas CBE 3.0 uas
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Stellar Effective Color <i>After color calibration for A and B</i>	REQ 8.0 uas CBE 6.0 uas
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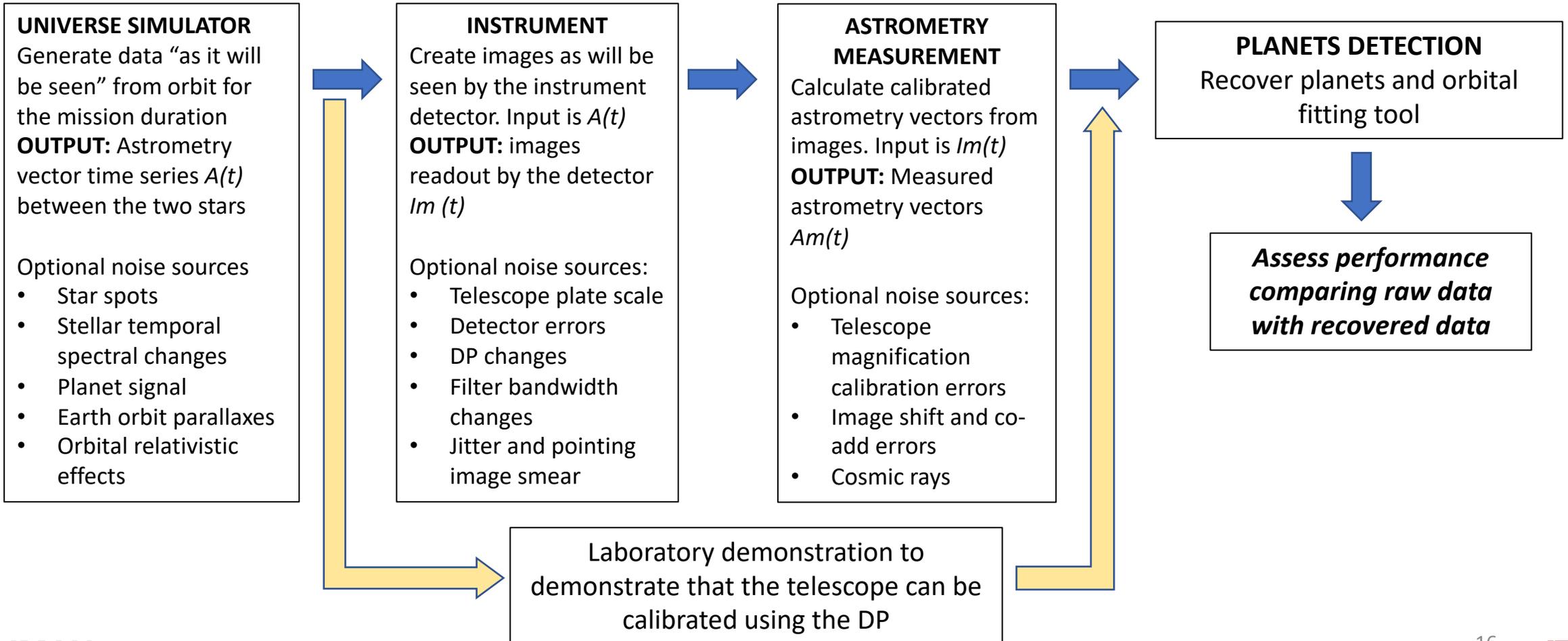
Exoplanet Photocenter bias <i>due to exoplanet reflected light or transits</i>
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Exozodi clumps

Instrument

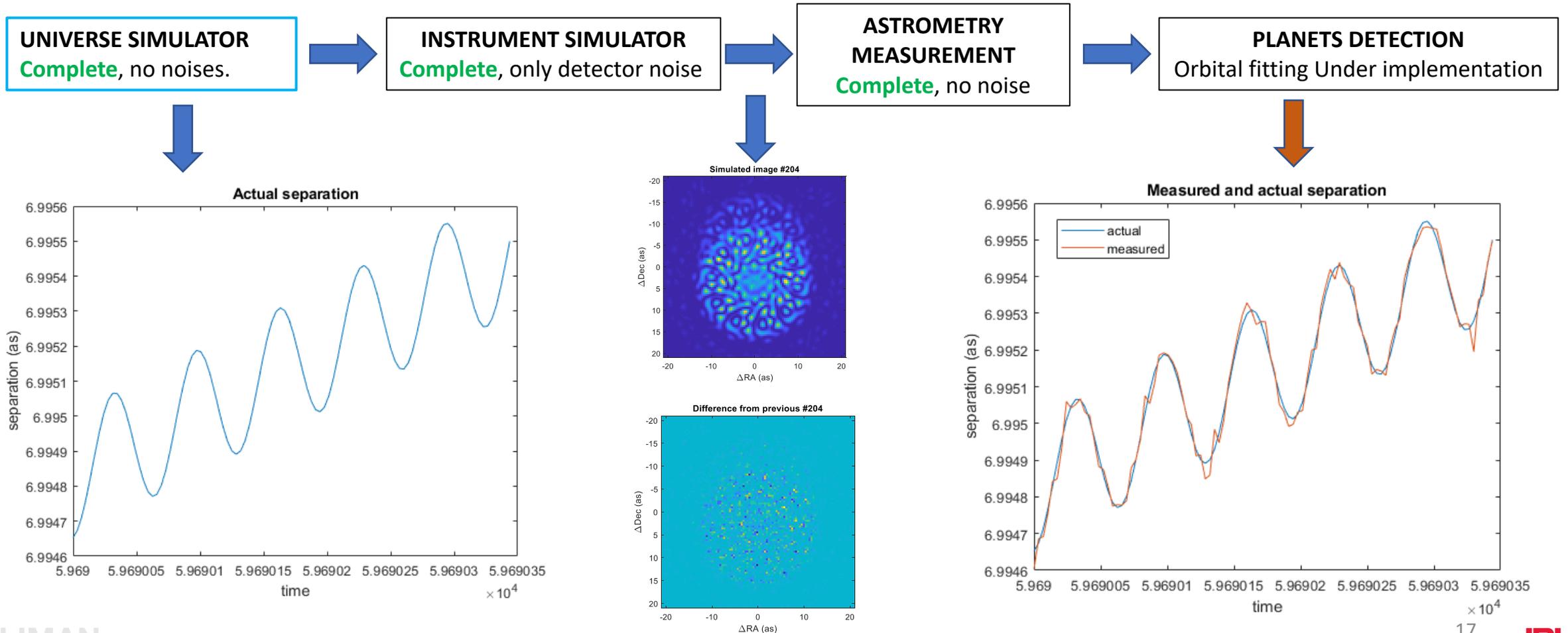
End-to-end simulation

We are simulating the entire process from observations to planet retrieval



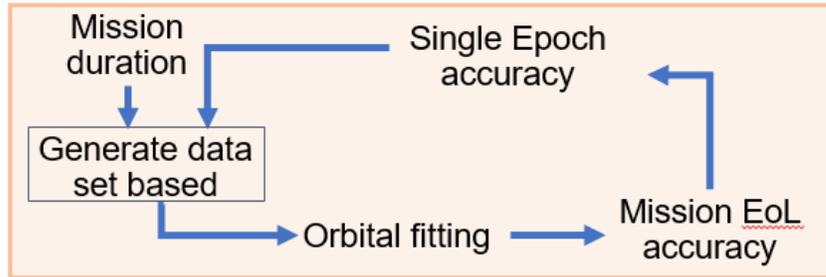
End-to-end simulation

Part of the simulator is already working



Orbital Fitting

Requirements derivation

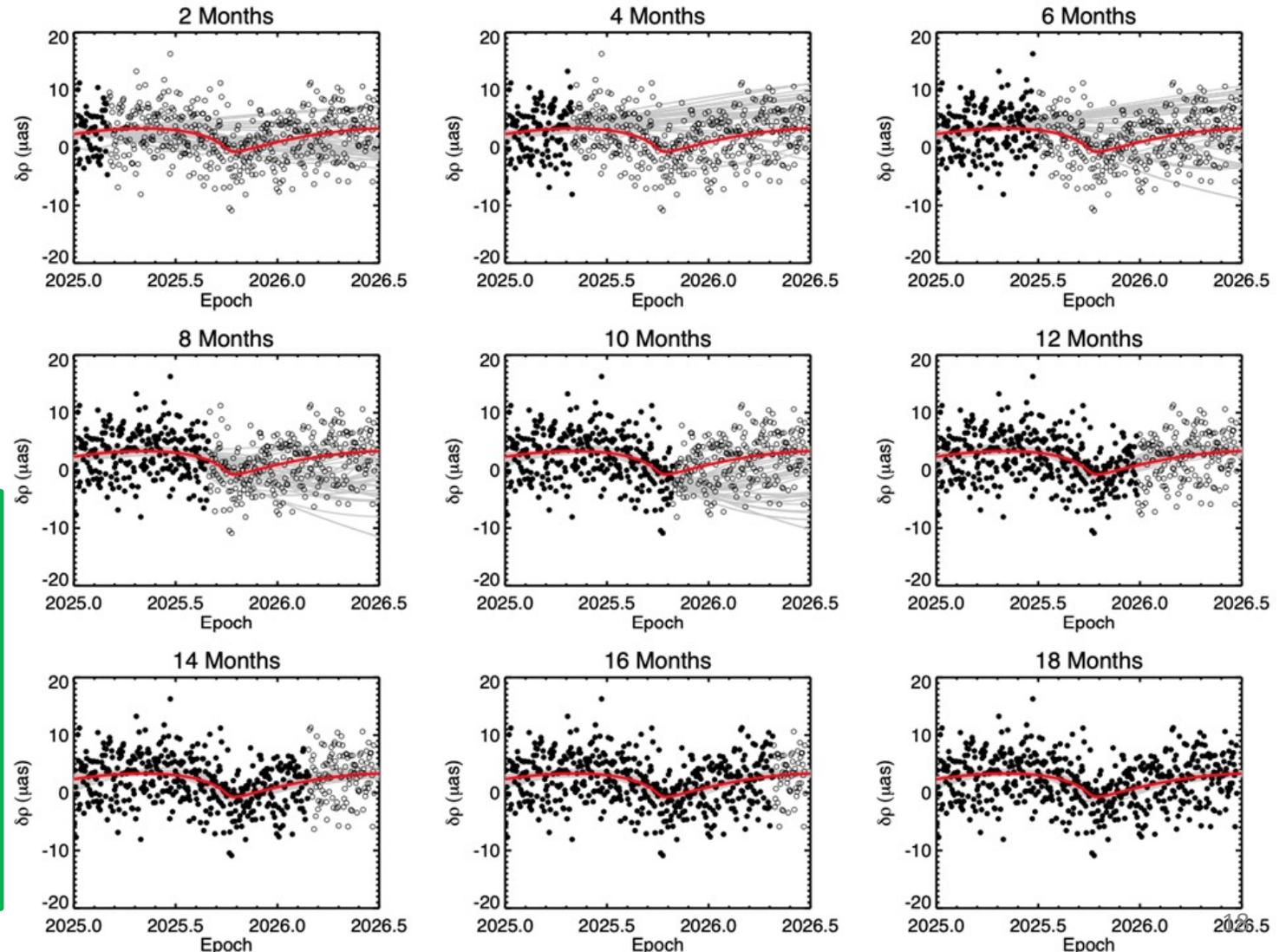


E. Nielsen
University of New Mexico

Anyone interested in doing a postdoc at NASA/JPL or thesis to work on these models?

- Adding RV
- Resolving host star degeneracy
- Blind spots

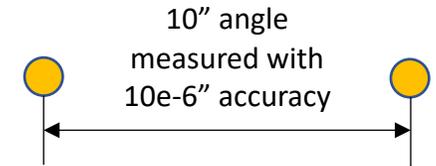
Talk to me, or email me at eduardo.Bendek@jpl.nasa.gov



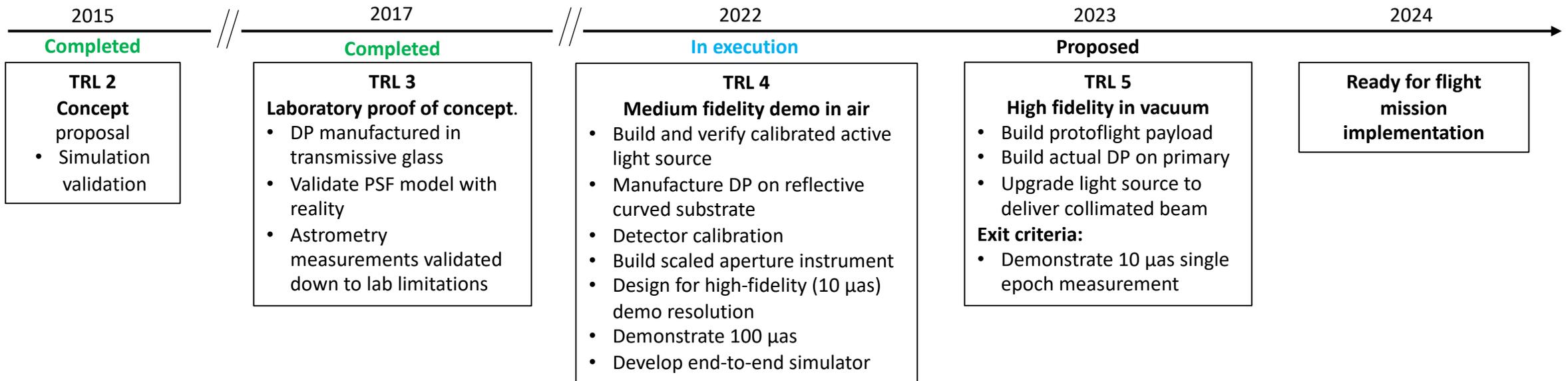
Technology demo roadmap

Goals

1. Demonstrate end-to-end exoplanet injection and recovery
2. Demonstrate single-epoch relative astrometry accuracy that enables the mission:
 - 10 μs for a 9.0 cm aperture for sources separated by 10" ($7 \lambda/D$) => 1 ppm signal ($7e-6 \lambda/D$)
 - Demonstrate manufacturability of the DP



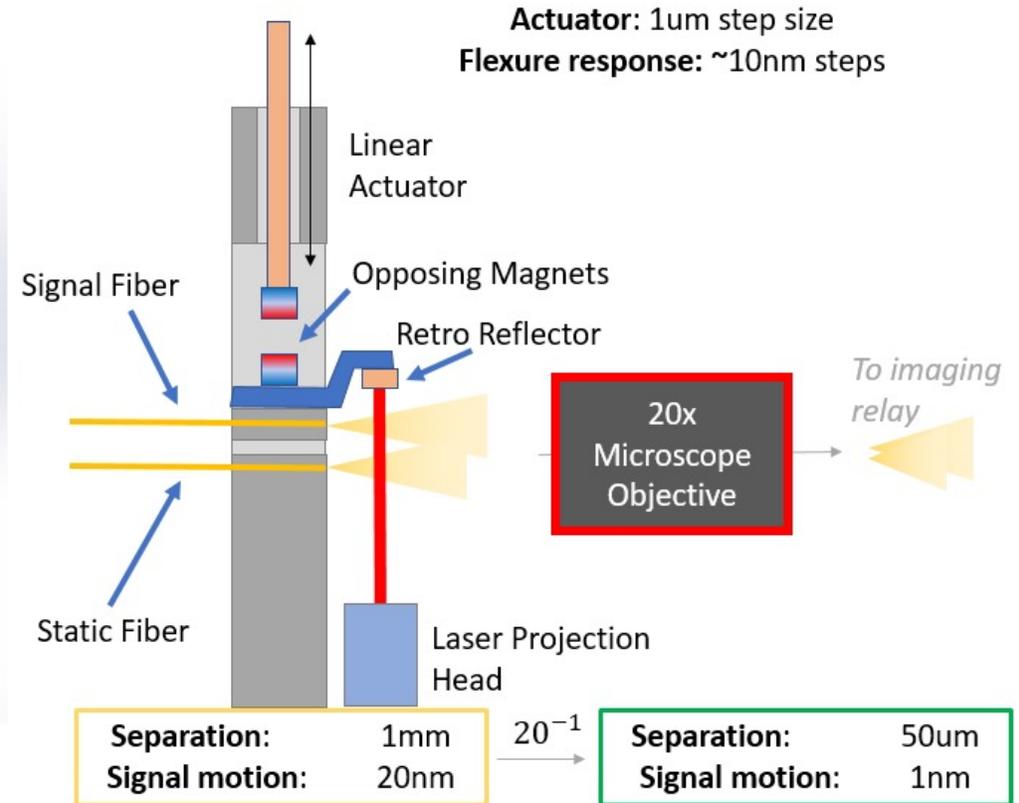
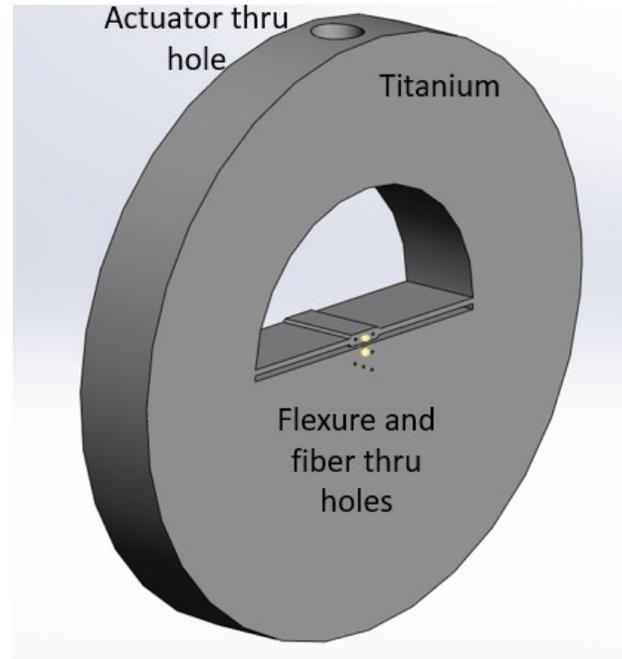
Roadmap



Laboratory demonstration

The universe simulator: An active star simulator

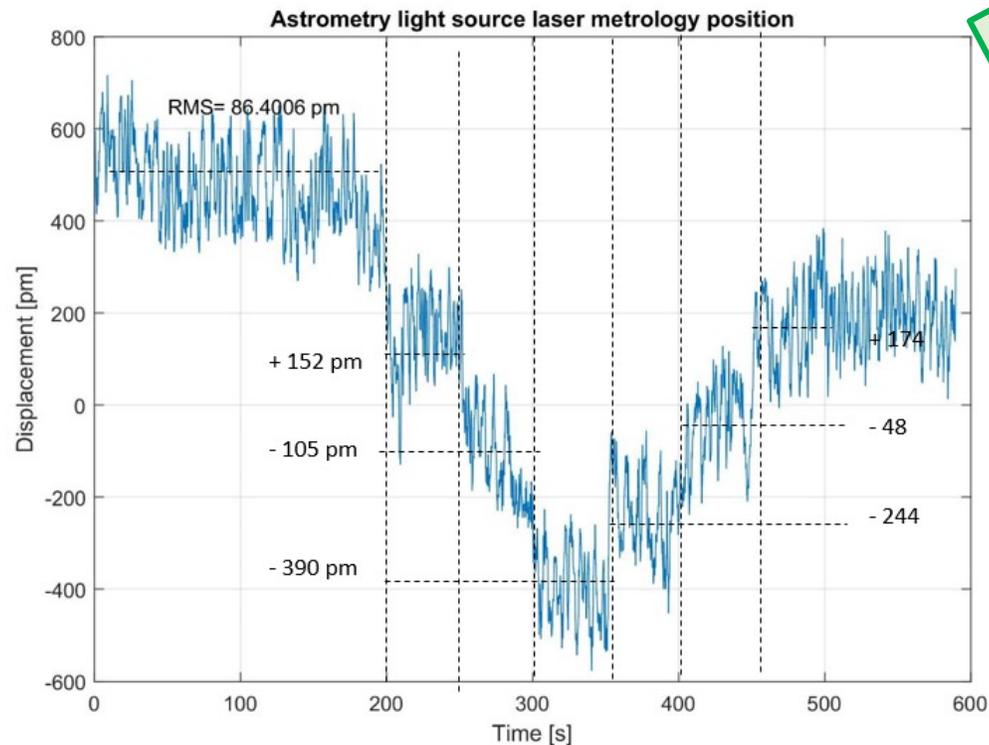
- Simulate Alpha Cen A and B and inject a 10 μ s signal (70 pm signal)
- Temporal modulation mitigate system drifts
- 50 μ m source separation to match the 7 λ /D angle on sky



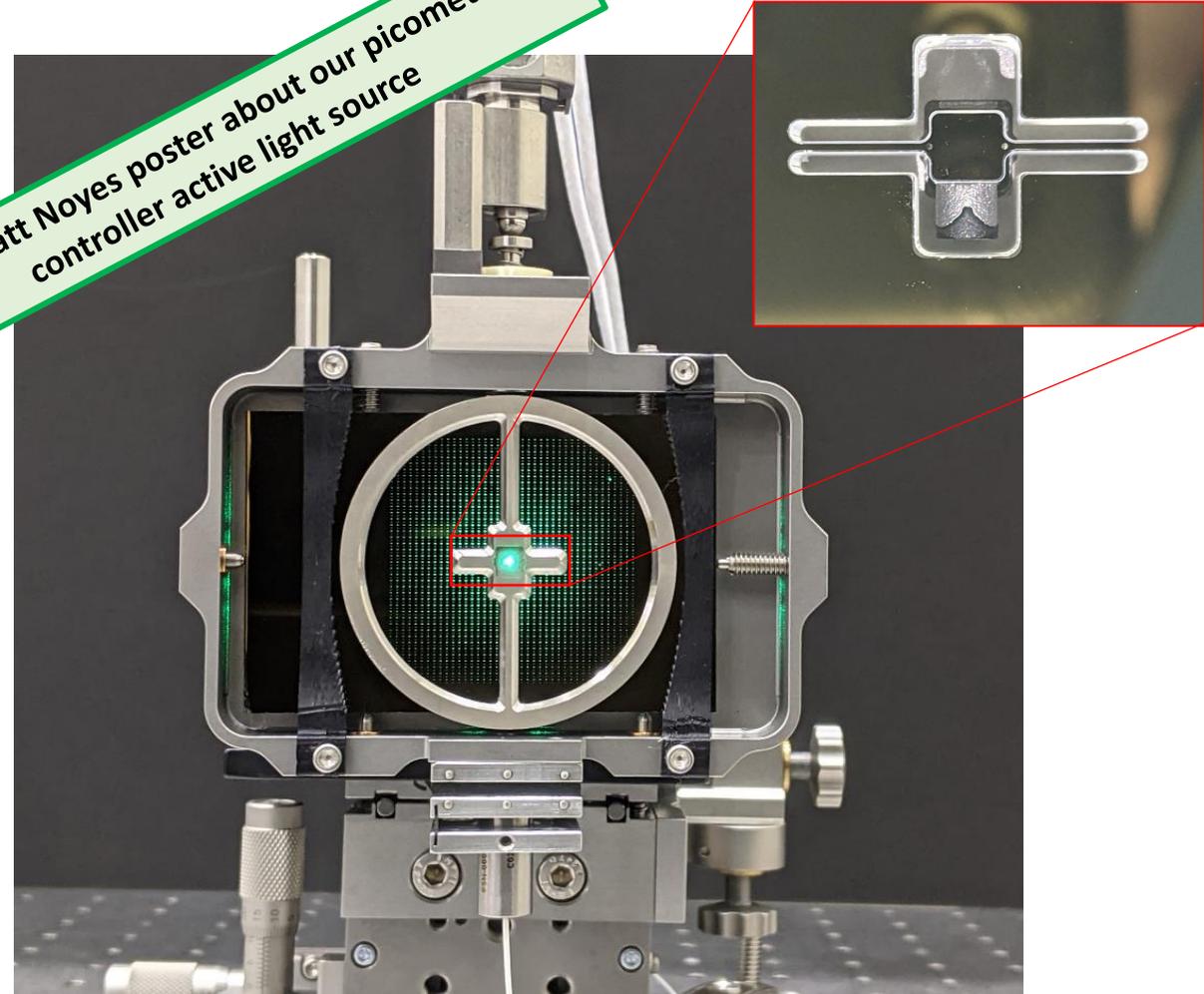
Laboratory demonstration

Heritage of APRA light source

- Measurable 200 pm step control using laser metrology
- 5x higher resolution than needed for TOLIMAN



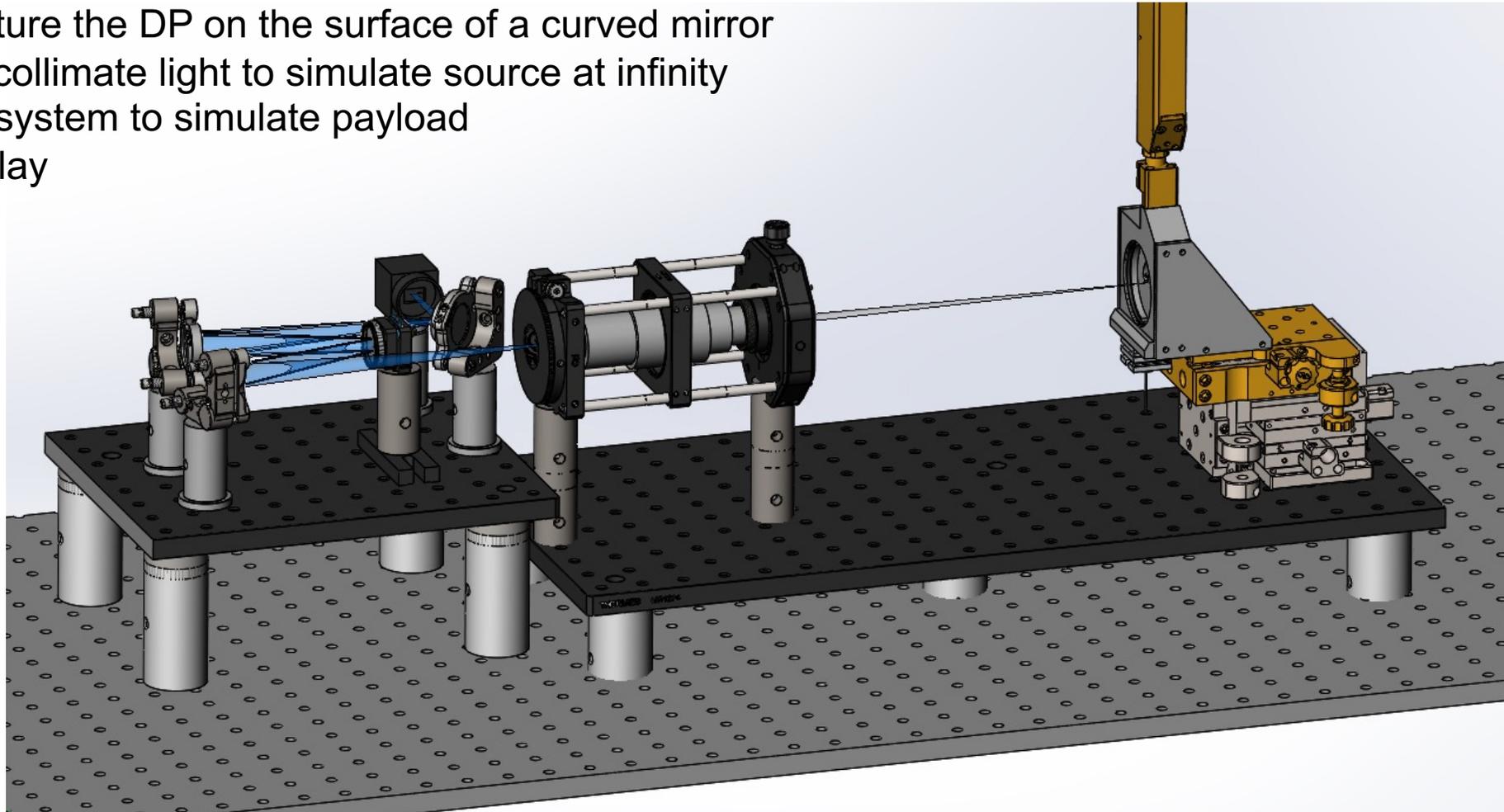
See Matt Noyes poster about our picometer controller active light source



Laboratory demonstration

The imaging system: Collimator and instrument payload

- Manufacture the DP on the surface of a curved mirror
- Need to collimate light to simulate source at infinity
- Imaging system to simulate payload
- Offner relay



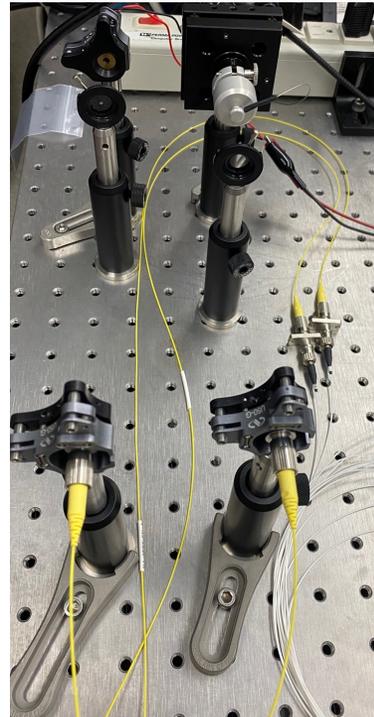
Laboratory demonstration is in preparation

Combination of current and previous investments

Source: **JNEXT**

Achievements:

1. Secured **lab space and optical bench**
2. Secured two white broadband **lasers** for source
3. Acquired and evaluated **CAMERA** candidate for flight
4. **Optical and mechanical** layout is almost finished.
5. Work with MDL ongoing for **DP manufacturing**



Binary Target Simulation

Injection of Finaium supercontinuum laser in one of the sources and narrow band laser of the other source to simulate a binary target



FLIR Camera intended for flight
(Board level for flight)

Same electronics as the Mars2020 EDL camera



Off-axis reflective optics

Next steps and conclusions

Science

- Finding the easiest targets to study exoplanets focuses us on nearby FGK
- Alpha Centauri is a unique target for its proximity and brightness

Technology

- Narrow angle relative astrometry is one of the most effective approaches to detect planets and measure masses around binaries
- Astrometry accuracy is limited by optical distortions that can be calibrated using a diffractive pupil

Implementation

- A cube sat with a 9 cm aperture is capable of collecting enough photons to perform the detection

Simulations: We are looking for people to help with modeling.